



December 31, 1963

Final Summary Report, | February 18, 1962 to December 31, 1963**INVESTIGATION OF THE SPACE STORABILITY  
OF PRESSURIZING GASES***Final... Report, Feb... 1963*

Prepared for:

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
WESTERN OPERATIONS OFFICE  
SANTA MONICA, CALIFORNIA( NASA CONTRACT NO. NAS7-105 )NASA CR-55841) OTS; #  
; SRI Project No. PSU-4000

Approved:

*R. F. Muraca*R. F. MURACA, ASSISTANT GENERAL MANAGER  
PHYSICAL SCIENCES RESEARCH

31 Dec. 1963 18p refs

Copy No. 137

## ABSTRACT

16714

A

Laboratory experiments indicate that a column of lithium borohydride powder will generate hydrogen on demand by reaction with controlled additions of water to an asbestos wick when cobaltous chloride is used as a catalyst and asbestos pulp is interspersed throughout to prevent formation of impermeable, glassy coatings of lithium metaborate.

A comparison is made of computed storage efficiencies for pressurized hydrogen and nitrogen in a system designed to generate hydrogen at about 1.5 atm from lithium borohydride with water added in a controlled fashion by flow through a porous plug. The maximum storage efficiency of such a hydrogen gas generating system, based on its use in an attitude-control system, approaches 27.7 lb-sec of impulse per pound of system weight for a mission requiring a total impulse of 10,000 lb-sec; hydrogen gas and nitrogen gas stored at 1500 psia have values of 16.6 and 33.9 lb-sec per lb, respectively. Thermal decomposition of lithium borohydride or other materials presents no advantage.

A UT#012

-R. F. Muraca

## **FINAL SUMMARY REPORT**

**SRI Project PSU-4000**

The Interim Report entitled "Investigation of the Space Storability of Pressurizing Gases," NASA Contract NAS7-105, dated August 1963 describes the major portion of work performed on this contract over the period February 19, 1962 to February 18, 1963.

As a part of this contract, a handbook was to be issued which would include information valuable to engineers designing pressurized-gas systems. A rough draft of the handbook was approved for publication by the program monitors (Messrs. Compitello and Porter). The published volume is entitled "Design Data for Pressurized Gas Systems," and dated November 1963; it is a loose-leaf binder containing 15 sections of data on various topics, including extensive tabulations and graphs of the physical and thermodynamic properties of pressurizing fluids.

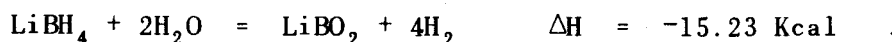
The work performed during the period February 19, 1963 to December 31, 1963 consisted largely of the preparation of material for the design data book, and needs no formal reporting here. The succeeding pages of this report summarize work performed on the evaluation of a candidate gas generating system suitable for attitude-control, pressurized-gas supply systems.

## STORAGE EFFICIENCY OF LITHIUM BOROHYDRIDE SYSTEMS

Some alternatives to simple pressurization for storage of gases to be used as working fluids for attitude control systems were given in Section VII of the Interim Report on this project (August 1963). Preliminary calculations indicated that if pressurized nitrogen is used for comparison there should be a weight advantage in the use of lithium borohydride as a source of gas for attitude control systems.

### INTRODUCTION

Lithium borohydride,  $\text{LiBH}_4$ , is available from commercial sources as a white to grayish microcrystalline powder or lumps. It is stable at room temperature, but the powder must be handled in air with great caution. Generally, its transfer is effected in atmospheres of dry nitrogen or argon (dry box). Water vapor in the atmosphere readily reacts with lithium borohydride:

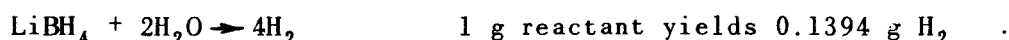
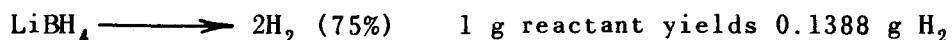


The heat of reaction may cause thermal decomposition and the material may deflagrate in air (especially the powder). Traces of water on combustible material like cellulose generally will react violently with  $\text{LiBH}_4$ ; as a result when  $\text{LiBH}_4$  powder comes into contact with materials like paper or cloth, spontaneous combustion is likely to occur. The properties of lithium borohydride are summarized in Table I.

Table I  
PROPERTIES OF LITHIUM BOROHYDRIDE

Formula:	$\text{LiBH}_4$
Formula Weight:	21.79
Structure:	ionic, $\text{Li}^+ \text{BH}_4^-$
Crystal Parameters:	orthorhombic, 6.81; 4.43; 7.17
Melting Point:	284°C (decomp.)
Heat of Formation:	44.1 Kcal/mole (Exo.)
Specific Heat:	0.84 cal/g/°C
Solubility:	soluble (with decomposition) in water and alcohols; 4 g/100 g ethyl ether; 21 g/100 g tetrahydrofuran

A cursory survey of the literature indicated that the mode of thermal decomposition is little understood and that only 75% of the hydrogen content of lithium borohydride is readily released by heat. The thermal decomposition is, of course, endothermic; thus the reaction could be readily used to supply hydrogen on command by application of heat. With only 75% recovery of hydrogen from the thermal decomposition of  $\text{LiBH}_4$ , the evolution of hydrogen by reaction with water is competitive:

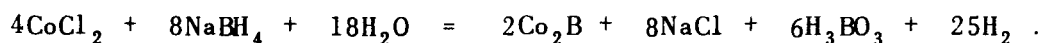


Further, it was believed simpler to be able to control the generation of hydrogen by metered additions of water than to control heat input to a relatively nonconductive mass of reactants. Thus, the investigation of the parameters governing the release of hydrogen from  $\text{LiBH}_4$  was begun.

#### REACTION OF WATER WITH $\text{LiBH}_4$

Available data on the reaction of water with lithium borohydride simply indicated that reaction takes place readily, but no information was available on the smoothness or completeness of reaction. On the other hand, the reaction of sodium borohydride with water has been studied in some detail by Schlesinger, *et al.*\*; the salient point to be made is that the hydrolytic reaction of sodium borohydride is slow, but that it can be accelerated by various substances and notably cobaltous chloride. The effect of the cobalt salt appears to be through route of formation of a black material of empirical composition  $\text{Co}_2\text{B}$  which is formed in the initial stages of the reaction of water with sodium borohydride, and this material serves to catalyze the subsequent reaction.

As a result of the studies, pellets containing 92.5%  $\text{NaBH}_4$  and 7.5%  $\text{CoCl}_2$  are offered on the market as "Hydropills" by Metal Hydrides, Inc.; the pills produce hydrogen gas readily upon contact with water. The formation of the catalyst appears to involve the reaction:



The pills may be used to replace conventional laboratory hydrogen generators.

---

\* Schlesinger, H. I., Brown, H. C., Finholt, A. S., Gilbreath, J. R., Ho Hoekstra, H. R., and Hyde, E. K., *J. Am. Chem. Soc.*, **75**, 215 (1953).

Lithium borohydride, as expected, also is hydrolyzed slowly by water; the hydrolysis is accelerated by cobalt salts analogously to the sodium derivative, and is of interest for possible use as a hydrogen generant in space applications since it offers a reduction in weight of reactants by about 22% over the sodium salt. The weight-volume relationships for the reaction of lithium borohydride with water are summarized in Table II.

Table II  
WEIGHT-VOLUME RELATIONSHIPS FOR THE REACTION:  
 $\text{LiBH}_4 + 2\text{H}_2\text{O} = \text{LiBO}_2 + 4\text{H}_2$

WEIGHT			VOLUME OF $\text{H}_2$ (STP)
$\text{LiBH}_4$	$\text{H}_2\text{O}$	$\text{H}_2$	
1.000 g	1.654 g	0.3699 g	4.115 liters
0.6048 g	1.000 g	0.2237 g	2.488 liters
2.703 g	4.470 g	1.000 g	11.12 liters
0.2430 g	0.4019 g	0.0899 g	1.000 liters
1.000 lb	1.654 lb	0.3699 lb	65.91 ft <sup>3</sup>
0.6048 lb	1.000 lb	0.2237 lb	39.86 ft <sup>3</sup>
2.703 lb	4.470 lb	1.000 lb	178.18 ft <sup>3</sup>
0.01517 lb	0.02697 lb	0.005612 lb	1.000 ft <sup>3</sup>

#### LABORATORY EXPERIMENTS—WATER

Pellets were prepared from a mixture of 5% anhydrous cobaltous chloride and 95%  $\text{LiBH}_4$  (Metal Hydrides, Inc.) in an inert atmosphere. It was found that only when excess water was added was 95-98% of the theoretical volume of hydrogen liberated from the pellets. Since the pellets were not prepared from material of the theoretical purity and under conditions in which moisture was strictly excluded, it was reasonable to assume that the yield of hydrogen was essentially quantitative. However, if only a small amount of water was added and the reaction allowed to proceed until hydrogen evolution ceased, further addition of water led to no, or at least a very slow, reaction; the sharply limited or inhibited rate of reaction was attributed to the presence of a tight "cake" of metaborate surrounding each pellet. It appears that the best method for obtaining quantitative generation of hydrogen from such pellets involves treating each pellet with sufficient water to dissolve the  $\text{LiBO}_2$  which is formed, or to add water to each pellet at such a rate that the concentrated solution of  $\text{LiBO}_2$  which initially forms is not dehydrated (by unreacted  $\text{LiBH}_4$ ) to a slowly-soluble glass. In any instance, however, it appears that the generation of hydrogen by reaction of water with pellets can not be controlled simply and, indeed may be eruptive. A reaction of this type obviously can not be reliably controlled in spacecrafts.

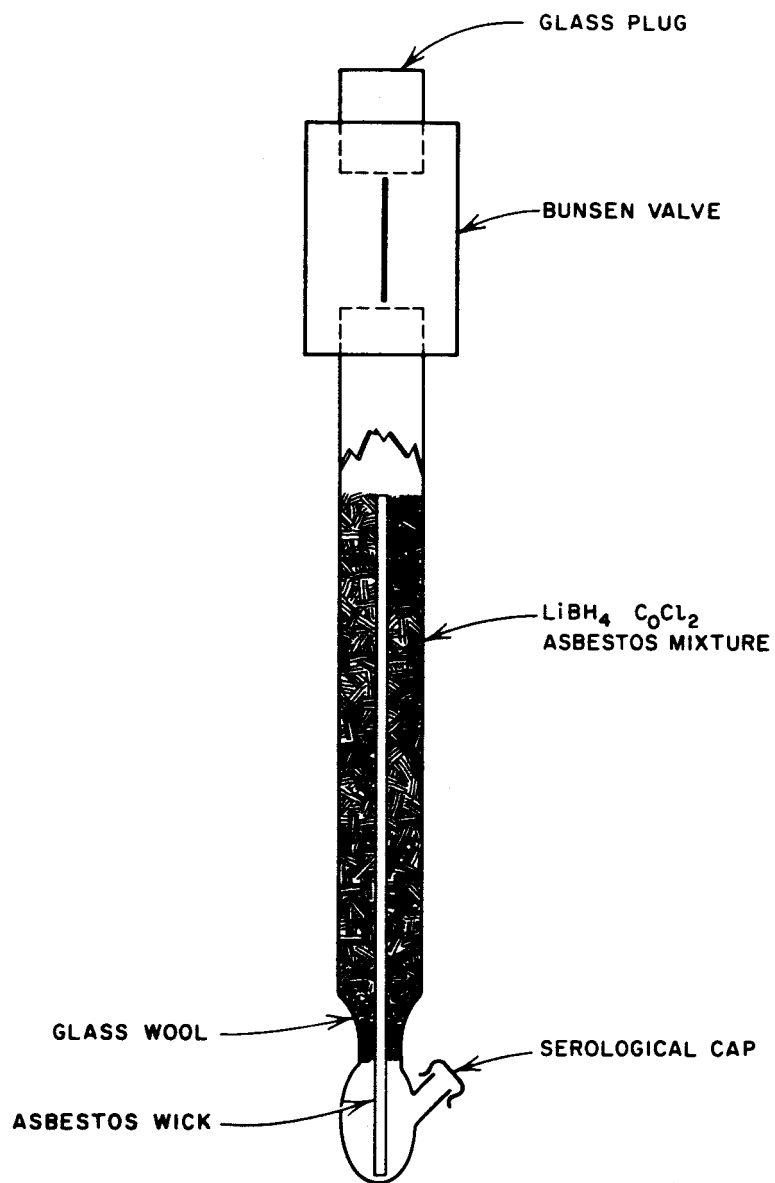
## LABORATORY EXPERIMENTS—WATER VAPOR

The reaction of water vapor with a powdered mixture of 95%  $\text{LiBH}_4$  and 5%  $\text{CoCl}_2$  was then investigated as a possible alternative route to the controlled formation of hydrogen from  $\text{LiBH}_4$ . The concept guiding the investigation was that water vapor can be generated easily in zero-g and its amount controlled by regulating the volume of water introduced into a system.

Preliminary experiments in which a  $\text{LiBH}_4$ - $\text{CoCl}_2$  powder mixture was allowed to have continuous access to water vapor revealed that an essentially quantitative yield of hydrogen could be obtained. In order to investigate the caking problem in intermittent operation, the simple apparatus shown in Figure 1 was constructed. (The filling configuration shown in this figure is the one finally adopted.) As indicated in the figure, the apparatus consists essentially of a reservoir with a side arm closed off by a serological cap, permitting injection of water *ad libitum* by an hypodermic syringe, a column in which powder can be placed, and a Bunsen valve to allow the escape of hydrogen and to inhibit entrance of the ambient atmosphere.

The particular arrangement shown in Figure 1 does not provide means for measurement of the evolved hydrogen; however, replacement of the Bunsen valve with rubber tubing leading to a measuring device made it possible to obtain quantitative measurements of the hydrogen released. The general course of the reaction could easily be followed in instances where the tube was packed with a mixture of  $\text{CoCl}_2$  and  $\text{LiBH}_4$  by noting the location of dark zones (black  $\text{Co}_2\text{B}$ ). Noting whether the dark material reacted with water provided a convenient check on the degree of decomposition.

In the first group of a series of experiments, water was introduced into the reservoir; the resulting water vapor slowly diffused into the powder column. It was found, as had been found before in experiments with water, that if a continuous supply of water vapor was available and the evolution of hydrogen was continuous, the generation of hydrogen proceeded smoothly and quantitatively. If, however, intermittent operation was attempted by adding only small increments of water to the reservoir, no further reaction could be obtained with subsequent additions of water, especially if considerable time elapsed between water additions.



RB-4000-28

FIG. 1 LABORATORY APPARATUS FOR STUDY OF REACTION  
OF WATER ON LITHIUM BOROHYDRIDE



In every case, where hydrogen evolution was inhibited, a well-defined, glassy cake was evident at the bottom of the column of powder. An experiment to avoid formation of the cake was then performed; a central wick of asbestos was used to transport water from the bottom reservoir into the powder mass. As might be expected, intermittent operation, though somewhat possible, was inhibited by formation of a cake around the wick.

The limited success with the use of an asbestos wick prompted a series of experiments in which the 95%  $\text{LiBH}_4$ -5%  $\text{CoCl}_2$  mixture was mixed with about 5% asbestos pulp and then tamped in the column around an asbestos wick; this is the assembly actually depicted in Figure 1. It was found that intermittent and essentially quantitative generation of hydrogen could be accomplished with this arrangement.

#### LABORATORY EXPERIMENTS—CONCLUSIONS

A mixture of 95%  $\text{LiBH}_4$ -5%  $\text{CoCl}_2$  will react smoothly and completely with liquid water or water vapor as long as the generation of hydrogen is continuous. However, special arrangements must be employed when intermittent generation of hydrogen is required, such as for application in space vehicles where the gas pressure must be held between specified limits.

The brief experimental feasibility study reported here was not designed to answer many obviously more detailed questions concerning hydrogen liberation. The desired intermittent operation was achieved by trying only one catalyst at one concentration (although the choice was reasonable from  $\text{NaBH}_4$  information), using a particular arrangement and relative weight of a specific fiber, and operating only at ambient temperature. Obviously, the interrelation among these variables and the rate of production and yield of hydrogen must be much better delineated before one can be assured with any accuracy that lithium borohydride can be part of a gas storage system for attitude control; but the information gained from the experiments permits design of a gas-generating system which has great possibilities for working successfully.

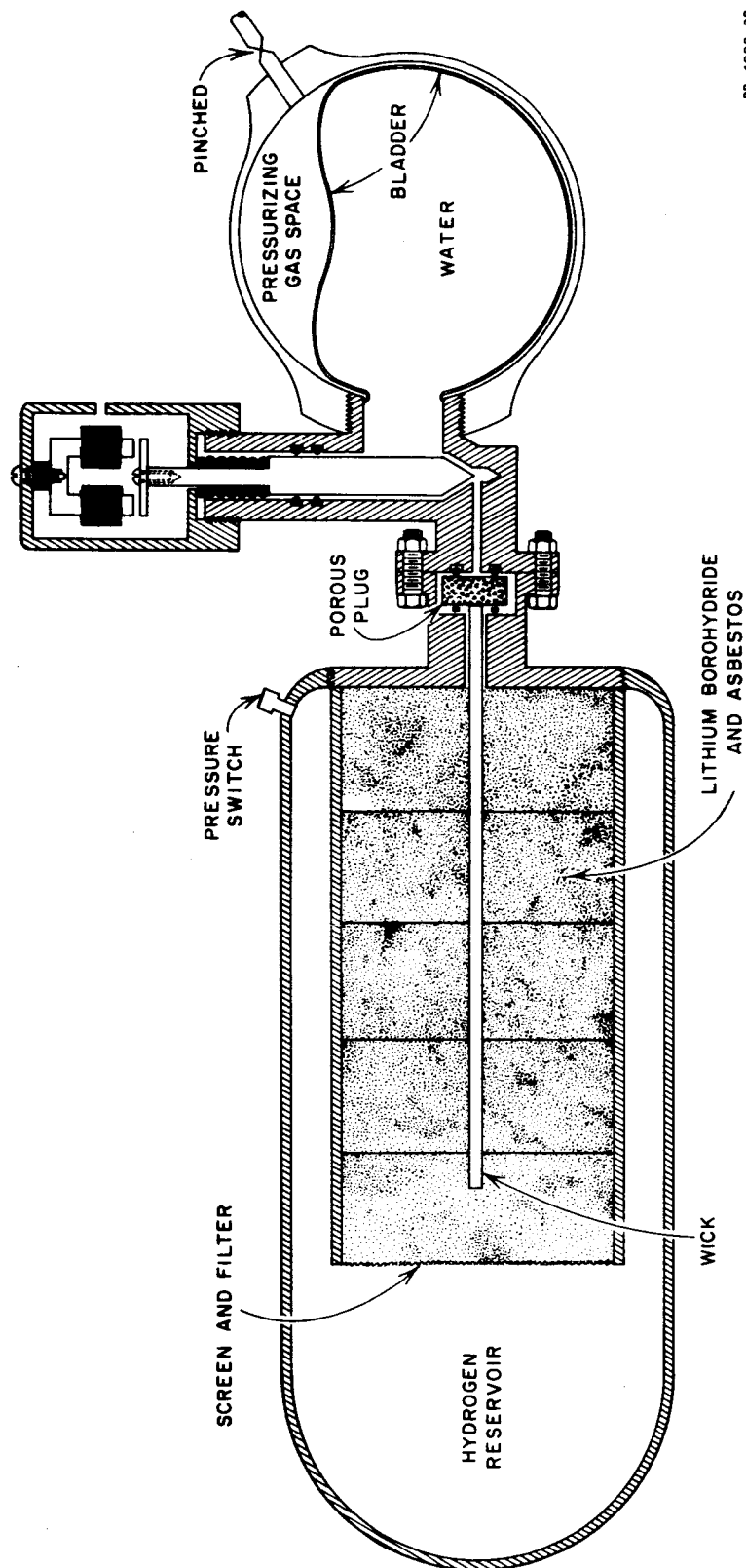
#### SYSTEM EVALUATION

The relative merits of gases such as nitrogen, hydrogen, propane, ammonia, etc., in pressurized systems used for attitude control are

generally assessed on the basis of the impulse obtainable per pound of system weight; when comparing liquefied gases to "permanent" gases, there are, for example, gains from reduced container weights because of lower storage pressures, and losses from the lower performance of the liquefied gases. Because the lithium borohydride-water combination as a source of pressurized hydrogen for attitude-control systems is completely different from the ordinary pressurized-gas system, its merits can only be discerned by a relatively detailed comparison of the anticipated over-all system weight with those of ordinary systems.

The configuration of what is considered to be a workable  $\text{LiBH}_4\text{-H}_2\text{O}$  hydrogen-generating system is indicated schematically in Figure 2. Referring to this figure, water is contained in an elastomeric bag wholly within a metal sphere; a pressurizing gas (nitrogen) occupies the ullage. The opening of the elastomer bag is connected to the inlet of a magnetically-actuated, spring-loaded valve; when actuated, the valve permits water to pass into a very narrow passageway which terminates at the face of a dense, porous plug. The water, under pressure, slowly permeates through the plug and is conducted by capillarity through asbestos fibers to the reaction zone in which is an intimate mixture of 4.5%  $\text{CoCl}_2$ , 90.5%  $\text{LiBH}_4$ , and 5.0% asbestos pulp contained in the tubular appurtenance indicated in Figure 2. Hydrogen evolved by the reaction escapes into the confining cylinder where it is stored under a pressure of 1-2 atmospheres; when the pressure in the cylinder rises over about 1.5 atmospheres, the pressure switch releases the magnetically-actuated valve; the water in the valve duct, porous plug, and asbestos wicks continues to react, but the pressure in the confining cylinder will not rise much over 2 atmospheres.

The use of a porous plug appears to offer a simple reliable method for controlling the flow of water into the reaction zone; since the flow of water through the plug is proportional to pressure and the pressure gradually diminishes as water is consumed, the rate of flow of water into the reaction zone is in a measure proportional to the requirements for smooth generation of hydrogen. This self-regulating feature is one of the important advantages of the system shown in Figure 2, and it can be brought to a high degree of development by a careful matching of the denseness and dimensions of the porous plug with the rate of pressure drop in the pressurizing sphere and the necessary rate of water flow for a given configuration of reaction mixture and asbestos wicking.



RB-4000-29

FIG. 2 DIAGRAM OF  $\text{LiBH}_4\text{-H}_2\text{O}$  HYDROGEN-GENERATING SYSTEM

It is evident from the above description that the configuration depicted in Figure 2 is very light in weight for it is essentially a low-pressure system; the highest pressure, less than 25 atmospheres, occurs in the sphere for pressurizing the water required for gas generation. Thus, the materials of construction can be plastics (as epoxy-glass) or light metals of extremely thin gage; the thinness of the metals will be governed largely by the techniques available for fabrication and, of course, by the need to withstand accelerative forces in the launch phase.

The comparison of the estimated weights for a 1500-psi hydrogen system, and the  $\text{LiBH}_4\text{-H}_2\text{O}$  system of Figure 2 are shown graphically in Figure 3; the data and details for the various systems are recorded in Tables III, IV, and V. For simplicity, the systems were computed for missions requiring total impulses of 100, 200, 500, 1,000, 5,000, and 10,000 lb-seconds and with 1,500 psia as the maximum storage pressure for the permanent gases. Titanium metal 6Al-4V (120,000 psi) was considered as most suitable for construction of all systems, although it is conceded that some advantages may be gained by using glass-epoxy storage vessels; however, the advantage was considered insufficient for the purposes of these preliminary comparisons.

The theoretical thicknesses of the spherical pressure vessels were computed by the well-known equation:

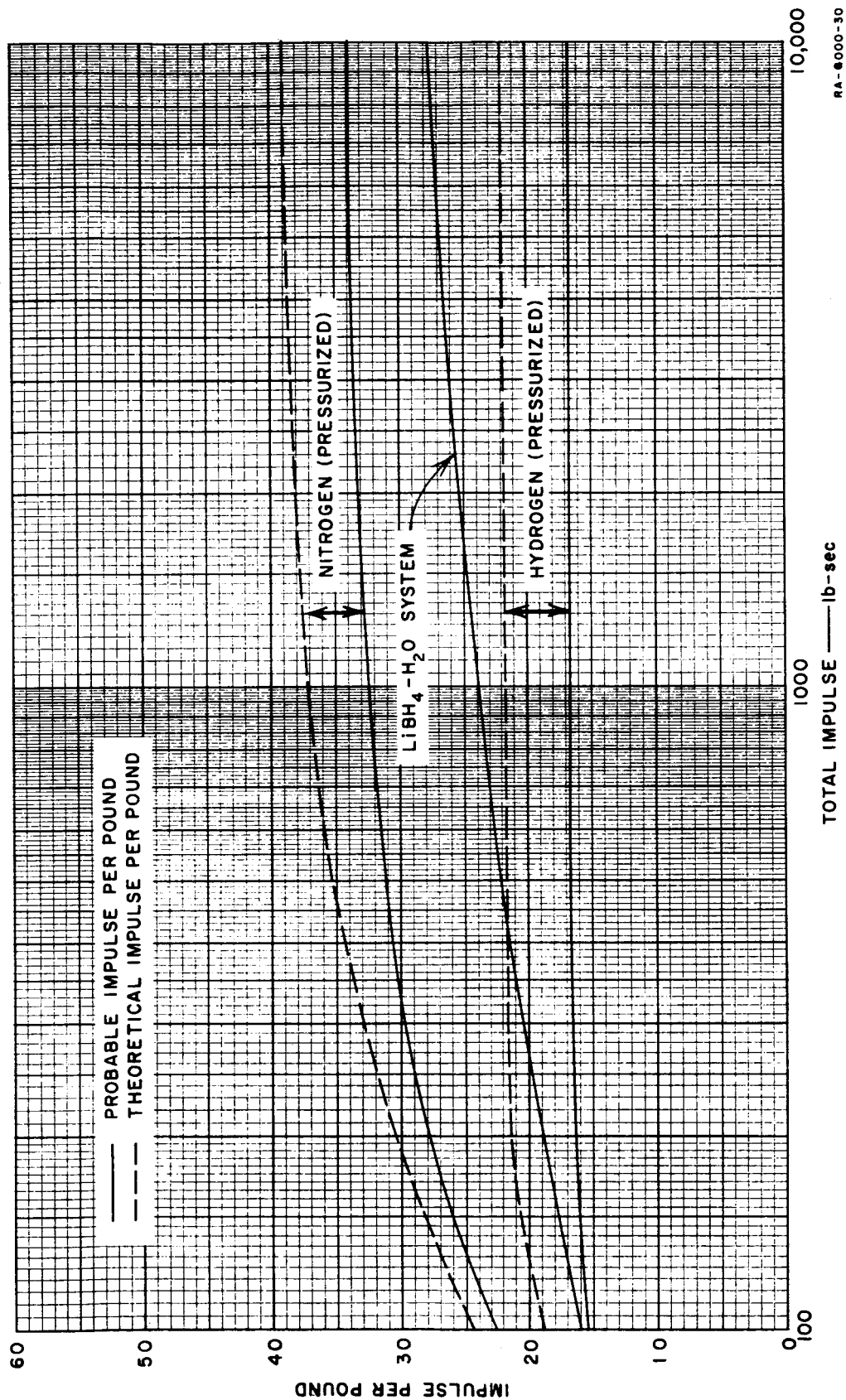
$$t = \frac{Pd}{4\sigma - P}$$

where  $P$  = pressure,  $d$  = inside diameter, and  $\sigma$  = the allowable tensile stress (120,000 psi for titanium).

No attempt was made to optimize the configuration of the gas storage container shown in Figure 2; for uniformity of computation, the following relationships were adopted:

$$L_1 = L_2 \quad \frac{L_1}{D_1} = \frac{6}{1} \quad \frac{D_2}{D_1} = \frac{1.5}{1}$$

The Volume,  $V$ , of the inner cylinder containing the  $\text{LiBH}_4\text{-CoCl}_2$ -asbestos mixture is 10% greater than the volume of the weight needed to provide



RA-6000-30

FIG. 3 EFFICIENCIES OF PRESSURIZED GAS STORAGE COMPARED WITH  $\text{LiBH}_4\text{-H}_2\text{O}$  GAS-GENERATING SYSTEM

Table III  
EFFICIENCY OF PRESSURIZED NITROGEN GAS STORAGE

IMPULSE* (lb-sec)	GAS WT (lb)	TANK DIAM† (ft)	THEO. TANK WT§ (lb)	PROBABLE TANK WT§ (lb)	THEO. IMPULSE (per lb)	PROBABLE IMPULSE (per lb)
100	1.45	0.730	2.60	2.97	24.7	22.6
200	2.91	0.922	3.69	4.44	30.3	27.2
500	7.27	1.26	6.91	8.79	35.3	31.1
1,000	14.5	1.58	12.4	16.1	37.2	32.6
5,000	72.7	2.93	58.6	94.0	38.0	33.0
10,000	145.	3.42	110.	147.	39.0	33.9

\* Based on nitrogen delivered at 1.5 atm and 538°R to a nozzle with an expansion ratio of 100; sp. ht. ratio = 1.40; calculated  $I_{sp}$  = 68.8 sec.

† Based on nitrogen storage at 1,500 psia at 538°R; density = 7.07 lb/ft<sup>3</sup>.

§ With added weights: reducing valve = 0.5 lb- control valve = 0.75 lb; fill port = 0.30 lb.

Table IV  
EFFICIENCY OF PRESSURIZED HYDROGEN GAS STORAGE

IMPULSE* (lb-sec)	GAS WT (lb)	TANK DIAM† (ft)	THEO. TANK WT (lb)	PROBABLE TANK WT (lb)	THEO. IMPULSE (per lb)	PROBABLE IMPULSE (per lb)
100	0.389	1.18	4.50	6.08	18.8	15.5
200	0.778	1.46	8.50	11.5	21.5	16.3
500	1.946	1.98	21.1	28.5	21.6	16.4
1,000	3.891	2.50	42.5	57.4	21.7	16.5
5,000	19.46	4.25	209.	282.	21.8	16.6
10,000	38.91	5.36	419.	566.	21.8	16.6

\* Based on hydrogen delivered at 1.5 atm and 538°R to a nozzle with an expansion ratio of 100; sp. ht. ratio = 1.40; calculated  $I_{sp}$  = 257 sec.

† Based on hydrogen storage at 1500 psia at 538°R; density = 0.482 lb/ft<sup>3</sup>.

Table V  
EFFICIENCY OF LITHIUM BOROHYDRIDE-WATER GENERATING SYSTEM

IMPULSE* (lb.-sec)	GAS WT (lb)	WATER WT (lb)	POWDER WT (lb)	POWDER VOL (ft <sup>3</sup> )	WATER VOL (ft <sup>3</sup> )	SPHERE DIAM (ft)	SPHERE WT† (lb)	BLADDER WT (lb)	N <sub>2</sub> GAS WT§ (lb)	TOTAL SPHERE WTΔ (lb)	D <sub>1</sub> (ft)	L <sub>1</sub> = L <sub>2</sub> (ft)	D <sub>2</sub> (ft)	TOTAL CYL. WTΔ (lb)	TOTAL SYSTEM WT# (lb)	THEO. IMPULSE (per lb)
100	0.389	1.739	1.162	0.067	0.028	0.415	0.249	0.025	0.007	2.30	0.250	1.50	0.375	1.64	6.30	15.9
200	0.778	3.479	2.324	0.133	0.056	0.530	0.407	0.039	0.013	4.50	0.314	1.88	0.471	2.58	10.6	18.9
500	1.946	8.696	5.811	0.333	0.139	0.705	0.719	0.072	0.033	10.9	0.427	2.56	0.641	4.78	22.7	22.0
1,000	3.891	17.39	11.62	0.665	0.279	0.890	1.15	0.116	0.066	21.5	0.537	3.22	0.806	7.56	41.9	23.9
5,000	19.46	86.96	58.11	3.33	1.39	1.53	3.39	0.335	0.372	105.	0.919	5.51	1.38	22.2	186.	26.9
10,000	38.91	173.9	116.2	6.65	2.79	1.92	6.40	0.530	0.744	209.	1.158	6.95	1.74	35.2	361.	27.7

\* Based on hydrogen delivered at 1.5 atm and 538°R to a nozzle with an expansion ratio of 100; sp. ht. ratio = 1.40; calculated  $I_{sp} = 257$  sec.

† Minimum wall thickness assumed = 0.02 inch.

§ Density = 1.78 lb/ft<sup>3</sup> at 25 atm.

Δ Plus 5% for miscellaneous additions.

# Plus magnetic valve = 0.75 lb; fill port = 0.15 lb; switch = 0.3 lb.

the required hydrogen; the bulk volume of the mixture was assumed to be 17.5 lb/ft<sup>3</sup>; thus:

$$1.10 (\text{Vol of mix}) = \text{Vol inner cylinder} = \frac{\pi D_1^2}{4} L_1 = V$$

$$\frac{6}{4} \pi D_1^3 = V = 4.7124 D_1^3$$

The configuration of the gas storage cylinder was assumed to be a cylindrical shell of length  $L_2$  and diameter  $D_2$  with hemispherical ends of radius  $D_2/2$ .

The bladder confining the water required for the mission (10% in excess) is a 10-mil elastomeric diaphragm of specific gravity 1.00 (wt = 0.052 lb/ft<sup>2</sup>). The full volume of water in the pressurizing sphere occupies 75% of the space in the sphere.

The pressurizing gas in the LiBH<sub>4</sub>-H<sub>2</sub>O system is nitrogen at 25 atmospheres and the volume is adjusted to give a final pressure of 3 atmospheres.

The pressure of the hydrogen gas collected in the storage reservoir fluctuates between 1 and 2 atmospheres; the powder mixture required was computed on the basis that hydrogen would be liberated quantitatively.

Probable titanium tank weights are 30-40% heavier than calculated from the formula given above; an average of 35% was used.

## CONCLUSIONS

As can be seen in Figure 3, the lithium borohydride-water system offers some improvement over pressurized hydrogen gas storage systems for attitude-control systems. A gain of about 50% is realizable over the probable efficiency of pressurized hydrogen, but it is about 18% less than anticipated from pressurized nitrogen gas systems.

No significant gain in storage efficiency can be realized from thermal decomposition of LiBH<sub>4</sub> since, as was shown above, the same weight of materials would have to be borne aloft, and the analysis summarized in Table V clearly indicates that the hardware weight is not the dominant factor determining storage efficiency. In fact, if no hardware were involved, the "bare storage efficiency" of LiBH<sub>4</sub> itself



would be 34.4 compared to 33.9 achievable in practice with nitrogen. Table VI summarizes "bare storage efficiencies" for various materials; these values clearly show the superiority of pressurized gases over hydrides. It is interesting to note that water, electrolyzed to "knallgas" has one of the highest "bare storage efficiencies;" however, water vapor itself has an  $I_{sp}$  of about 95 (see "Design Data for Pressurized Gas Systems," November 1963).

Table VI  
BARE STORAGE EFFICIENCIES\* OF VARIOUS MATERIALS  
FOR MISSIONS OF 10,000 lb-sec TOTAL IMPULSE

MATERIAL	"BARE STORAGE EFFICIENCY"	REMARKS
$\text{LiBH}_4$	34.4	Thermal decomposition to hydrogen
$\text{LiH}$	32.6	Thermal decomposition to hydrogen
$\text{BeH}_2$	46.9	Thermal decomposition to hydrogen; existence of compound is questionable
$\text{B}_2\text{H}_6$	56.1	Gaseous decomposition at $T > 300^\circ\text{C}$ to $3\text{H}_2$
$\text{H}_2\text{O}$	85.9	Electrolysis to $\text{O}_2$ and $2\text{H}_2$
$\text{NH}_3$	88.3	Decomposition to $\text{N}_2$ and $3\text{H}_2$

\* "Bare storage efficiency" means the impulse per pound of material obtainable when gas liberated from it by an appropriate method is expanded through a nozzle of 100:1 expansion ratio at an input pressure of 1.5 atm and a temperature of  $540^\circ\text{R}$ .

## DISTRIBUTION LIST

ORGANIZATION	NO. OF COPIES	ORGANIZATION	NO. OF COPIES
NASA Western Operations Office 150 Pico Boulevard Santa Monica, California Office of Technical Information Contracting Officer Patent Office	1 1 1	Marshall Space Flight Center Huntsville, Alabama	2
NASA Lewis Research Center 21000 Brookpark Road Cleveland 35, Ohio Office of Technical Information Contracting Officer Patent Office	1 1 1	Manned Spacecraft Center Houston, Texas	2
NASA Marshall Space Flight Center Huntsville, Alabama Office of Technical Information, M-MS-IPC Contracting Officer, M-P&C-C Patent Office, M-PAT	1 1 1	Advanced Research Projects Agency Pentagon, Room 3D154 Washington 25, D.C.	1
NASA HEADQUARTERS Washington 25, D.C. Contracting Officer, BRA Patent Office, AGP	1 1	Aeronautical Systems Division Air Force Systems Command Wright-Patterson Air Force Base, Ohio	1
Mr. Henry Burlage, Jr. Chief, Liquid Propulsion Systems, RPL	4	Air Force Missile Development Center Holloman Air Force Base, New Mexico	1
Mr. A. O. Tischler Assistant Director for Propulsion, MLP	1	Air Force Missile Test Center Patrick Air Force Base, Florida	1
TECHNICAL MANAGER Mr. Richard N. Porter Liquid Propulsion Section Propulsion Division Jet Propulsion Laboratory 4800 Oak Grove Drive Pasadena, California	1	Air Force Systems Command, Dyna-Soar Air Force Unit Post Office Los Angeles 45, California	1
Scientific and Technical Information Facility NASA Representative, Code CRT P.O. Box 5700 Bethesda, Maryland 20014	25	Army Ordnance Missile Command Redstone Arsenal, Alabama	1
Ames Research Center Moffett Field, California	2	Arnold Engineering Development Center A.E.O.R. Tullahoma, Tennessee	1
Goddard Space Flight Center Greenbelt, Maryland	2	Bureau of Naval Weapons Department of the Navy Washington 25, D.C.	1
Jet Propulsion Laboratory California Institute of Technology 4800 Oak Grove Drive Pasadena, California	2	Central Intelligence Agency 2430 E Street, N.W. Washington 25, D.C.	1
Langley Research Center Langley Field, Virginia	2	Defense Documentation Center Headquarters Cameron Station, Building 5 Alexandria, Virginia 22314 Attn: TISIA	1
Lewis Research Center 21000 Brookpark Road Cleveland 35, Ohio	2	Headquarters, United States Air Force Washington 25, D.C.	1
		Picatinny Arsenal Dover, New Jersey	1
		Rocket Research Laboratories Edwards Air Force Base, California	1
		U.S. Naval Ordnance Test Station China Lake, California	1
		U.S. Atomic Energy Commission Technical Information Services Box 62 Oak Ridge, Tennessee	1

# DISTRIBUTION LIST

	ORGANIZATION	NO. OF COPIES	ORGANIZATION	NO. OF COPIES
Re	CPIA			
As	Chemical Propellant Information Agency	1	Fairchild Stratos Corporation	1
De	Johns Hopkins University		Aircraft Missiles Division	
Pr	Applied Physics Laboratory		Hagerstown, Maryland	
	8621 Georgia Avenue			
Re	Silver Spring, Maryland		General Electric Company	1
Ti			Missile and Space Vehicle Department	
Da	Aerojet-General Corporation	1	Box 8555	
	P.O. Box 296		Philadelphia, Pennsylvania	
R	Azusa, California			
F			General Electric Company	1
L	Aerojet-General Corporation	1	Rocket Propulsion Units	
	P.O. Box 1947		Building 300	
R	Sacramento 9, California		Cincinnati 15, Ohio	
D				
6	Aeronutronic	1	Grumman Aircraft Engineering Corporation	1
C	A Division of Ford Motor Company		Bethpage, Long Island, New York	
	Ford Road			
S	Newport Beach, California		Kidde Aero-Space Division	1
9			Walter Kidde and Company, Inc.	
E	Aerospace Corporation	1	675 Main Street	
	2400 East El Segundo Boulevard		Belleville 9, New Jersey	
S	El Segundo, California			
I			Lockheed Aircraft Corporation	1
I	Arthur D. Little, Inc.	1	Missile and Space Division	
	Acorn Park		Sunnyvale, California	
S	Cambridge 40, Massachusetts			
I			Lockheed California Company	1
	Astropower, Inc., Subsidiary of	1	10445 Glen Oaks Boulevard	
	Douglas Aircraft Company, Inc.		Pacoima, California	
	2968 Randolph Avenue			
	Costa Mesa, California		Lockheed Propulsion Company	1
			P.O. Box 111	
	Astrosystems, Inc.	1	Redlands, California	
	82 Naylor Avenue			
	Livingston, New Jersey		Marquardt Corporation	1
			16555 Saticoy Street	
	Atlantic Research Corporation	1	Box 2013 - South Annex	
	Edsall Road and Shirley Highway		Van Nuys, California	
	Alexandria, Virginia			
			Martin Division	1
	Beech Aircraft Corporation	1	Martin Marietta Corporation	
	Boulder Facility		Baltimore 3, Maryland	
	Box 631			
	Boulder, Colorado		Martin Denver Division	1
			Martin Marietta Corporation	
	Bell Aerosystems Company	1	Denver, Colorado	
	P.O. Box 1			
	Buffalo 5, New York		McDonnell Aircraft Corporation	1
			P.O. Box 6101	
	Bendix Systems Division	1	Lambert Field, Missouri	
	Bendix Corporation			
	Ann Arbor, Michigan		North American Aviation, Inc.	1
			Space and Information Systems Division	
	Boeing Company	1	Downey, California	
	P.O. Box 3707			
	Seattle 24, Washington		Northrop Corporation	1
			1001 East Broadway	
	Convair (Astronautics)	1	Hawthorne, California	
	Division of General Dynamics Corporation			
	P.O. Box 2672		Pratt & Whitney Aircraft Corporation	1
	San Diego 12, California		Florida Research & Development Center	
			West Palm Beach, Florida	
	Curtiss-Wright Corporation	1		
	Wright Aeronautical Division		Philco Corporation	1
	Wood-ridge, New Jersey		Western Development Laboratories	
			3825 Fabian Way	
	Douglas Aircraft Company, Inc.	1	Palo Alto, California	
	Missile and Space Systems Division			
	3000 Ocean Park Boulevard			
	Santa Monica, California			

**STANFORD  
RESEARCH  
INSTITUTE**

**MENLO PARK  
CALIFORNIA**

## **Regional Offices and Laboratories**

### **Southern California Laboratories**

820 Mission Street  
South Pasadena, California

### **Washington Office**

808-17th Street, N.W.  
Washington 6, D.C.

### **New York Office**

270 Park Avenue, Room 1770  
New York 17, New York

### **Detroit Office**

1025 East Maple Road  
Birmingham, Michigan

### **European Office**

Pelikanstrasse 37  
Zurich 1, Switzerland

### **Japan Office**

c/o Nomura Securities Co., Ltd.  
1-1 Nihonbashidori, Chuo-ku  
Tokyo, Japan

## **Representatives**

### **Toronto, Ontario, Canada**

Cyril A. Ing  
Room 710, 67 Yonge St.  
Toronto 1, Ontario, Canada

### **Milan, Italy**

Lorenzo Franceschini  
Via Macedonio Melloni, 49  
Milano, Italy